

# Chapter 6

## State Estimation

### Part 2

#### 6.2 Sensors for State Estimation



# Outline

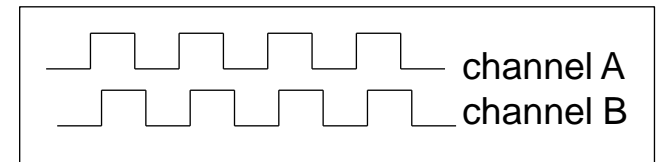
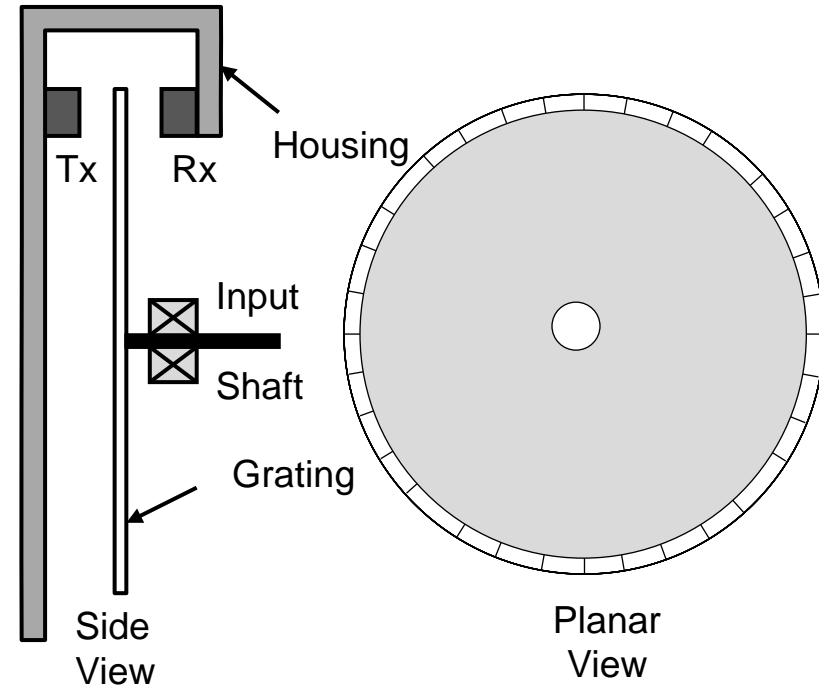
- 6.2 Sensors for State Estimation
  - 6.2.1 Articulation Sensors
  - 6.2.2 Ambient Field Sensors
  - 6.2.3 Inertial Frames of Reference
  - 6.2.4 Inertial Sensors
  - Summary

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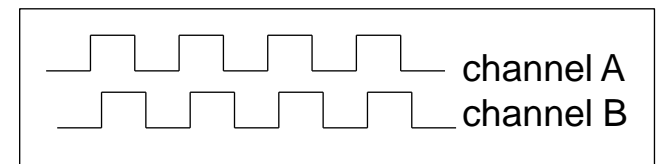
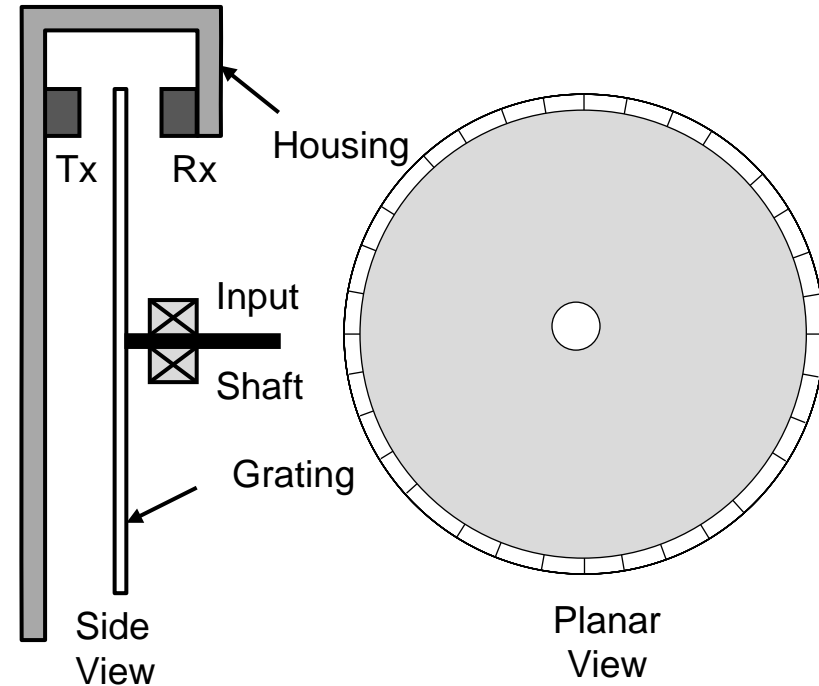
# 6.2.1.1 Optical Encoders

- Standard equipment on mobile robots. Used for wheel rotations, pan-tilt heads, etc.
  - Wheel slip issues mean **super high accuracy encoders don't make sense** (for wheels).
- A photodetector (photodiode) and phototransmitter (laser diode) are lined up to **look through the holes** in a grating.



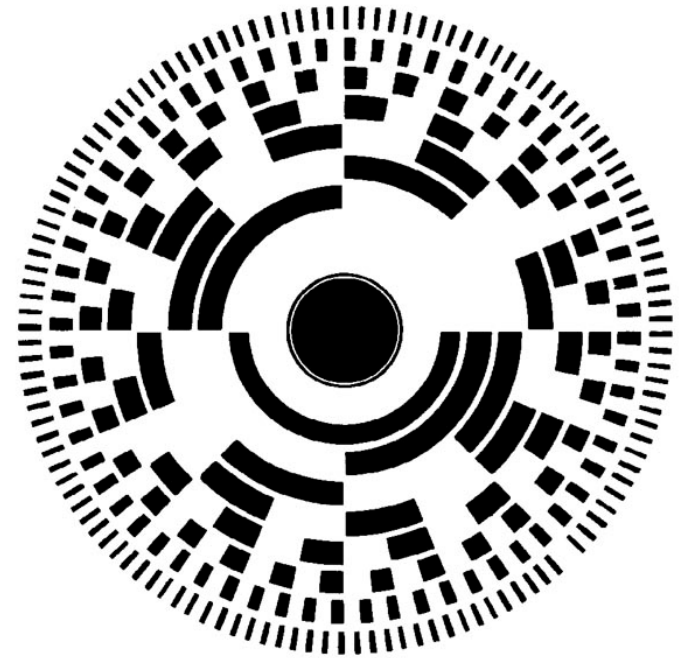
# 6.2.1.1 Optical Encoders

- Interrupted light source generates a series of **pulses** which are **counted**.
- A second Tx/Rx pair can generate a second set of pulses 90 degree out of phase with first.
  - Permits **sensing direction** of motion.
- 1000 to 10000 **counts per revolution** are typical.
  - Gear ratio and backlash sometimes affects output resolution.



## 6.2.1.2 Absolute Optical Encoders

- Relative encoders (like the above) **lose count when shut off**.
- Few solutions:
  - Drive physically to a **home position** after power is applied.
  - **n** concentric circles of gratings encoding binary numbers (opposite) and **n** receivers.
- Up to 26 bits precision is available!



**Absolute Encoder**  
(No need to count pulses now)

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# Field (Radiation) Sensors

- Moving in a field of known characteristics is **equivalent to having a map**.
- Cheap, low tech sensors for measuring attitude on the earth such as compasses and inclinometers **rely on the earth's fields** (magnetic and gravitational).
- This same principle is used in radio navigation but the field is generated **artificially**.
- All vision-based navigation uses the same principle.
- Attitude can be computed from the relative orientations of the vehicle body and two known non-collinear vectors.
  - Why Two?
  - Two are required because one does not constrain rotation about its direction.

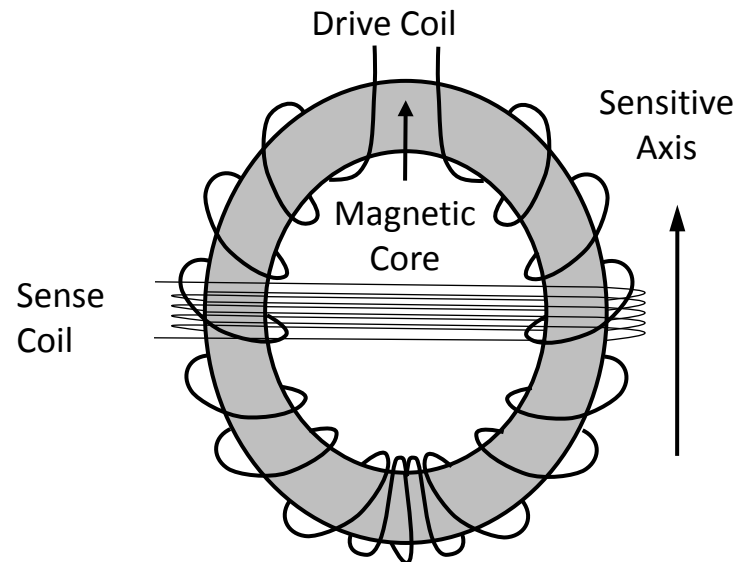


## 6.2.1.1 Compasses and Magnetometers

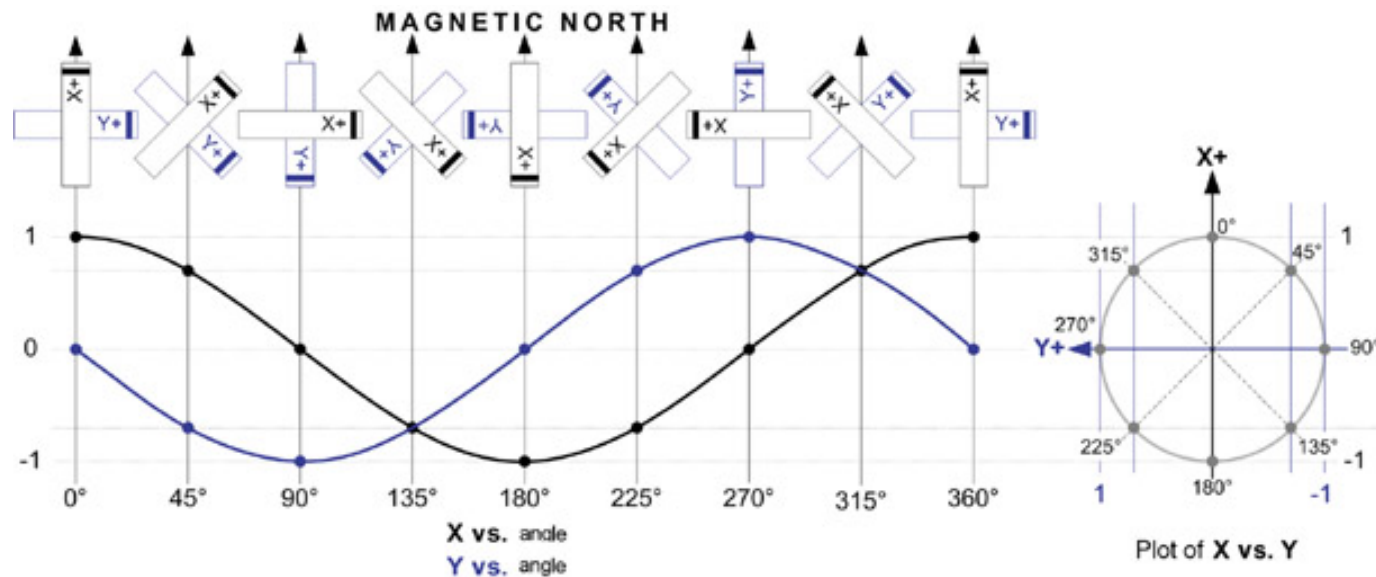
- **Compass** = a device that measures orientation with respect to the Earth's poles.
- **Gyrocompasses** have been used on ships
  - sense the true direction of the earth's spin axis.
- Only the **magnetic compass** is used on mobile robots.
  - sense earth's magnetic field as an orientation reference
- Often occur on mobile robots in the form of a (fluxgate) **magnetometer**.
  - sense the magnitude of the ambient magnetic field in one or more directions.

# Fluxgate Magnetometer

- Compare the drive-coil current needed to saturate the core in one direction as opposed to the opposite direction.
- Output the difference
  - Which is equal to external field magnitude.
- Saturation detected as nonlinearity.



# (2 axis) Magnetometer Response



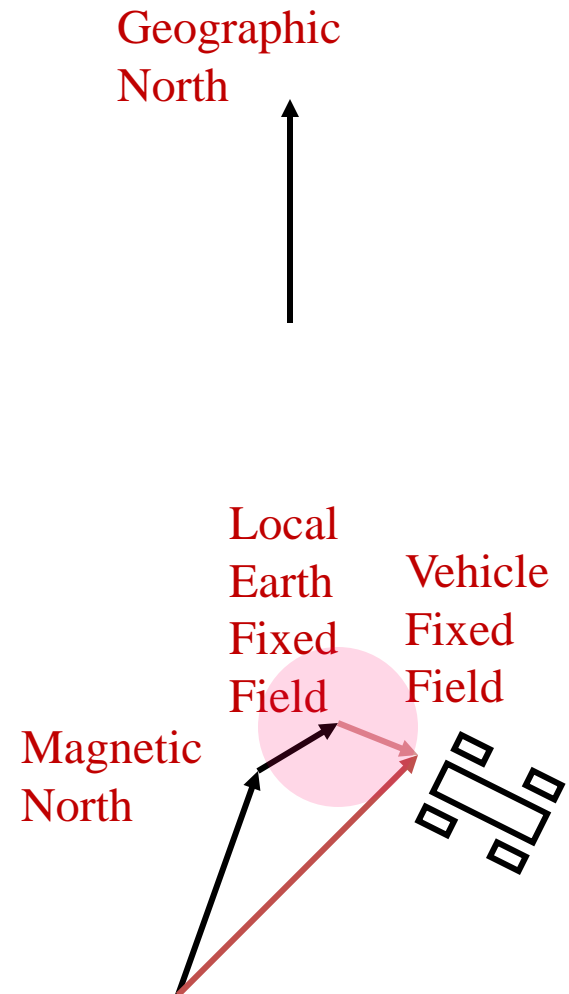
- $\text{Theta} = \text{atan2}(y,x)$
- Real devices often have slow response.
- Their alignment on the vehicle also needs to be calibrated.

# Earth's magnetic field: **Variation**

- Flow of liquid ferromagnetic material in Earth's core causes a very very weak field.
- Mariners used to rely on it and have charted all its many quirks.
  - Not aligned with geographic north (direction to earth's spin axis). Known as variation.
  - Magnetic north pole is in Canada, magnetic south pole is off coast of Australia.
  - Entire east coast of US is 15 degrees east variation.

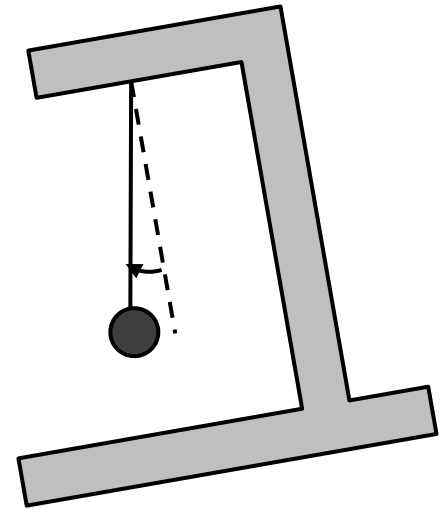
# Local Field: Deviation

- Some metals have magnetic permeability which redirects the ambient field.
- Permanent and electromagnets create fields which redirect the ambient field.
- Some of these things are fixed to the robot and some are fixed to the ground.
  - Some may be fixed to other robots.
- Can calibrate the vehicle field out by spinning in a circle and measuring difference from ground truth rotation.



## 6.2.2.2 Inclinometers

- Measure direction of specific force.
- Often work on identical principles as pendulous accelerometers.
  - Measure angle of proof mass deflection.
  - Cheap devices have bandwidths on the order of 2 Hz.
  - Rebalance devices are faster.
- Ideal solution for mobile robots that don't move ;)
- Why?



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## 6.2.3 Inertial Frames

- Simply, a frame where Newton's laws hold.
- Don't those laws hold everywhere in the universe?
  - Nope.
  - $F = ma$  **in an inertial frame** of reference.
- Remember the moving observer stuff? There were two accelerations measured. Which is right?
  - **Both** are.
  - But **the  $a$  times the  $m$  only equals the  $F$**  if the observer is in an inertial frame.
  - Otherwise the motion seems inconsistent with the  $F$  due to **apparent** other forces.



## 6.2.3 Inertial Frames

- A frame which is:
  - Far removed from any massive object (no gravitation), or...
  - Under no net gravitational force, or...
  - Free to move under the influence of gravitation (freefall)
- Third case subsumes the other two.
- Examples include:
  - Intergalactic space (far removed)
  - Center of the earth (no net  $g$ 's)
  - Spacecraft in orbit (why are you “weightless” if you are being held in orbit by the earth’s gravity?)
  - Aircraft in freefall (NASA trains astronauts in these)
- Anything that **rotates** wrt the stars is **not** inertial.

## 6.2.3.1 Apparent Forces

- Reverse our classical acceleration transform...

$$\vec{a}_o^m = \vec{a}_o^f - (\vec{a}_m^f + 2\vec{\omega}_m^f \times \vec{v}_o^m + \vec{\alpha}_m^f \times \vec{r}_o^m + \vec{\omega}_m^f \times [\vec{\omega}_m^f \times \vec{r}_o^m])$$

- If the **f** frame is inertial, only the  $\vec{a}_0^f$  term can be **explained by applied forces**.
- Yet, the moving observer sees motion that indicates other forces are being applied.
  - What “force” pushes you against the wall of a car in a turn?
  - The other terms are said to be caused by **apparent** forces.
  - 4 terms inside the brackets above are, in order, Einstein, Coriolis, Euler, Centrifugal

## 6.2.3.2 Noninertial Earth Fixed Frame of Reference

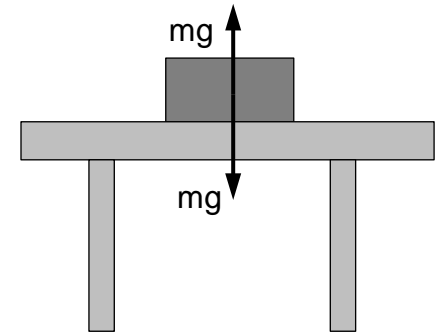
- Earth-fixed observer says table reaction **equals**  $mg$ .

$$\sum \vec{F} = \vec{R} + m\vec{g} = 0$$

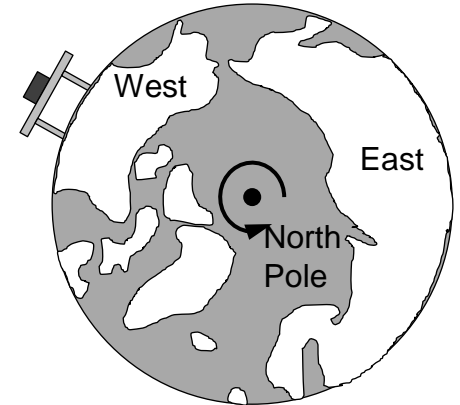
- Inertial observer says table reaction is **slightly less** – explains circular motion.

$$\sum \vec{F} = \vec{R} + \vec{G} = m\vec{a}$$

- Therefore !  $m\vec{g} = \vec{G} - m\vec{a}$
- Gravity ( $g$ ) is more than gravitation ( $G$ ).
  - Contains apparent forces !
- BTW:  $g$  depends on latitude



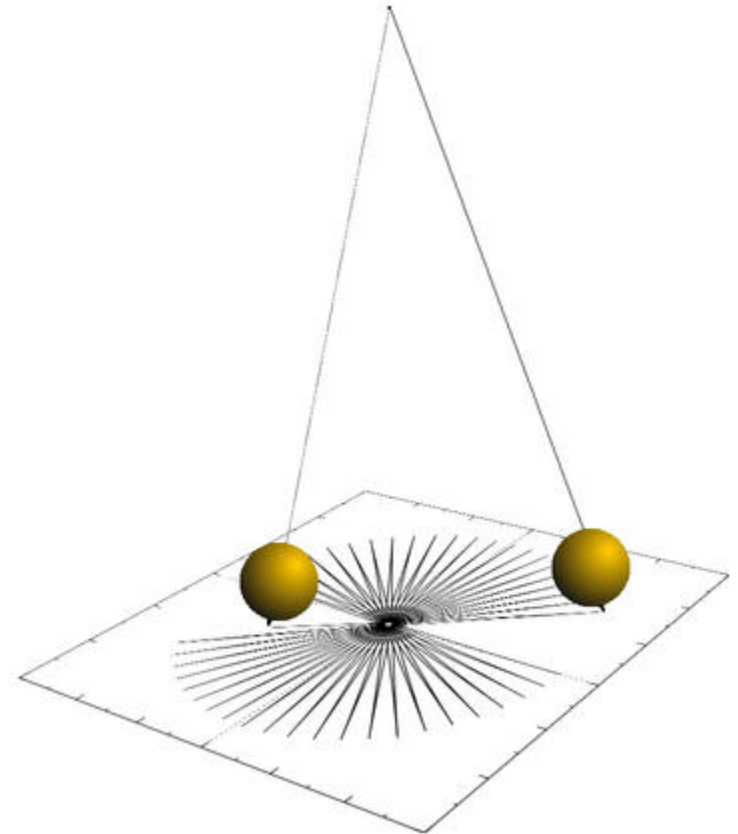
Earth-Fixed Observer



Inertial Observer

# 6.2.3.2.1 Effect of Earth's Rotation

- The surface of the earth is not an inertial frame.
- Almost every  $f=ma$  you wrote in high school was **wrong**.
- Instruments can detect this. Some examples:
  - Gyroscope at equator pointing “up” rotates once per day.
  - Foucault pendulum in UN building in New York.
  - Objects don't fall in straight lines.
  - Gyrocompass oscillates due to earth's rotation.



Foucault Pendulum

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    - 6.2.4.2 Accelerometers
    - 6.2.4.3 Gyroscopes
    - 6.2.4.1 Inertial Measurement Units
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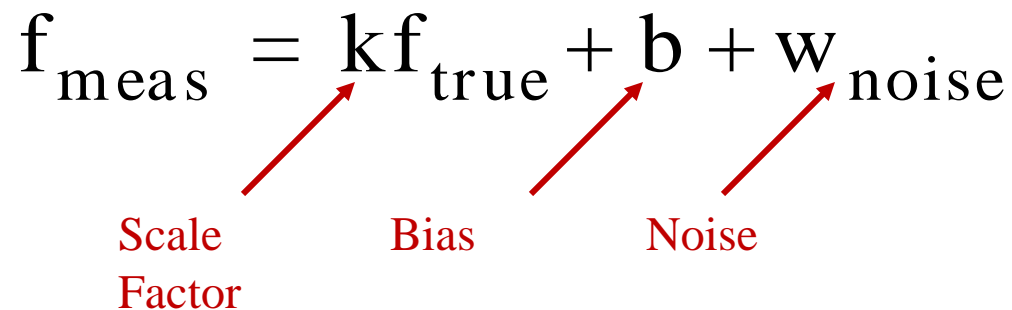
## 6.2.4 Inertial Sensors

- Higher tech sensors such as accelerometers and gyroscopes are based on inertial principles.
- Accelerometers and inclinometers can each do the other's job:
  - accelerometers can provide attitude (w.r.t gravity) when acceleration is known.
  - inclinometers can provide acceleration when attitude is known.

## 6.2.4.1 Inertial Sensor Performance

- Devices are sensitive to temperature, shock, and vibration.
- A simple model of performance is:

$$f_{\text{meas}} = k f_{\text{true}} + b + w_{\text{noise}}$$

  
Scale Factor                  Bias                  Noise

- Because autocalibration is the rule, it's the **stability of bias and scale factors that matters most.**

# Measuring Noise

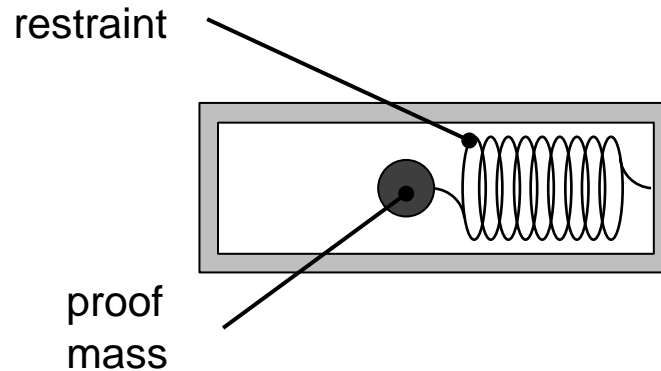
- Can be characterized in terms of the **spectral amplitude**:
  - For gyros get “density” in  $\text{deg}/\text{sec}/\text{Hz}^{1/2}$  or  $(\text{deg}/\text{sec})^2/\text{Hz}$
- **Angle Random Walk (ARW)**: Describes the error magnitude expected from integrating a noisy rate measurement.
  - After removing effects of bias and scale.
  - Slope of the integrated velocity noise process curve in  $\text{deg}/\text{root}(\text{hr})$  (stdv) or  $\text{deg}^2/\text{hr}$  (variance).
- **Velocity Random Walk**: Integral of an assumed acceleration white noise term.



# Outline

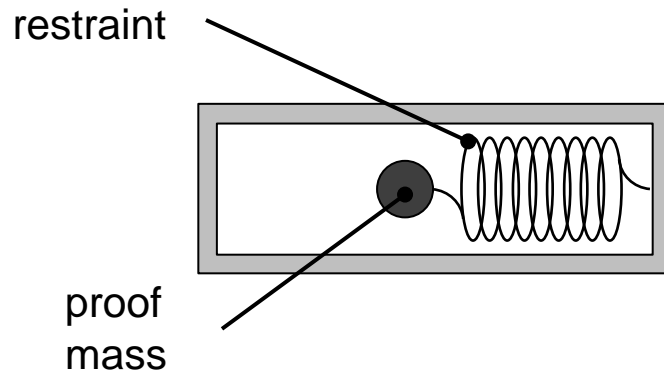
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## 6.2.4.2 Accelerometers



- Accelerometers are typically divided into two classes, depending upon their intended use.
- **Guidance accelerometers** are those intended for use in measuring the steady state accelerations of rigid bodies. One might use a guidance accelerometer for measuring the acceleration of an automobile.
- **Vibratory or seismic accelerometers** are those intended to measure sinusoidal accelerations. They are used to measure vibrations in applications as varied as structural testing, and earthquake and tsunami detection.

# 6.2.4.2.1 Operating Principle



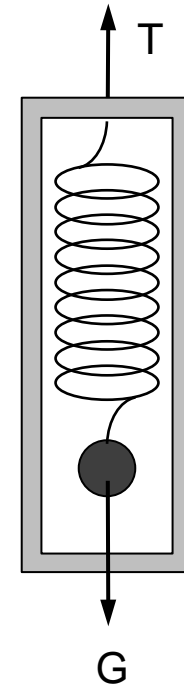
Deformable objects will deform **under acceleration or when transmitting forces**

- All accelerometers operate on the same principle, that of measuring the relative displacement of a small mass, called a proof or seismic mass, constrained within an accelerating case.
  - An accelerometer is therefore a physical manifestation of Einstein's famous box from general relativity.
- The restraint
  - may be elastic, viscous, or electromagnetic.
  - is a transducer that returns a signal proportional to the displacement of the proof mass.
- Typically, the mass is only allowed a single degree of freedom which may be either linear or rotary.

## 6.2.4.2.2 Specific Force

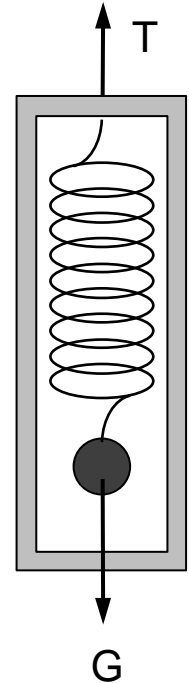
- Accelerometers measure **specific force, not acceleration** and there is nothing we can do about it.
- Case 1: When the **case is accelerated**, the spring will deflect - allowing us to measure the acceleration.
- Case 2: When the **case is placed upright on a table**, it will measure the influence of gravitation because the spring will deflect with the weight of the mass.

“Specific” means per unit mass.



# 6.2.4.2.2 Specific Force (Subtleties)

- The deflection of the elastic restraint is directly influenced by the **tension in the spring** - not by the mass's acceleration.
- From Newton's Second Law; the **inertial acceleration** of the mass ( $a^i$ ) is **proportional to the total force** applied to the mass - in this case, the gravitational force ( $F_g$ ) and the elastic restraining force ( $T$ ).



- Specific force:

$$\vec{T} + \vec{G} = m\vec{a}^i$$

$$\frac{\vec{T}}{m} = \vec{a}^i - \frac{\vec{G}}{m}$$

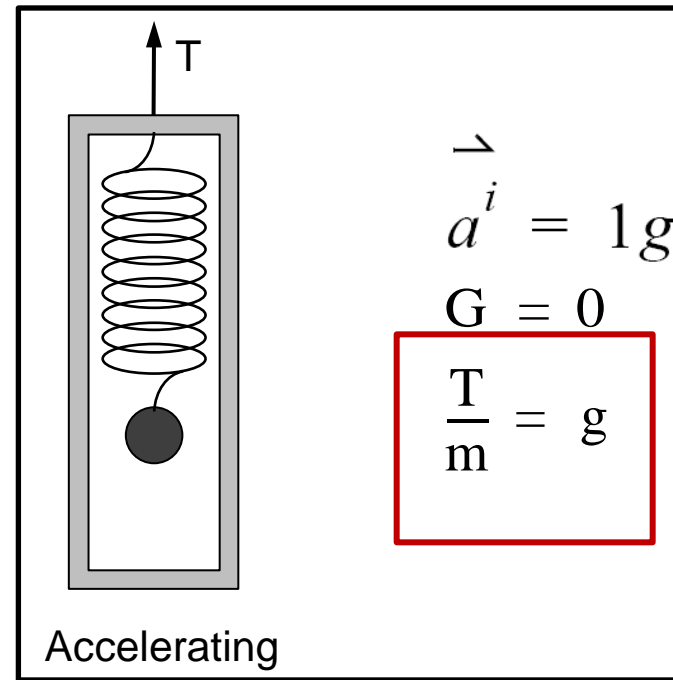
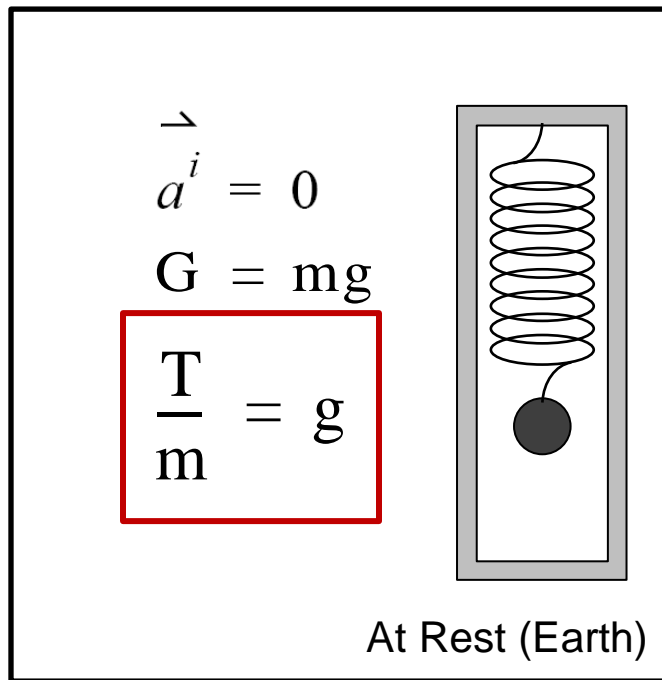
Acceleration proportional to net force, not  $T$ .

Spring deflection is proportional NOT to  $a$ , but to  $a-g$ .

## 6.2.4.2.2 Specific Force

(Equivalence of Gravitation and Acceleration)

- An accelerometer cannot tell the difference between these two effects.



# 6.2.4.2.2 Specific Force

(Accelerometers as Dynamic Systems)

- All accelerometers have some stiffness and some damping.
  - Modulating enables tuning the response.
  - Guidance accelerometers are **low pass**.
  - Vibratory accelerometers are **bandpass**.
- Devices with **rebalance loops** actively oppose motion and report the effort (current) required to do so.
  - These generally have much faster response times.

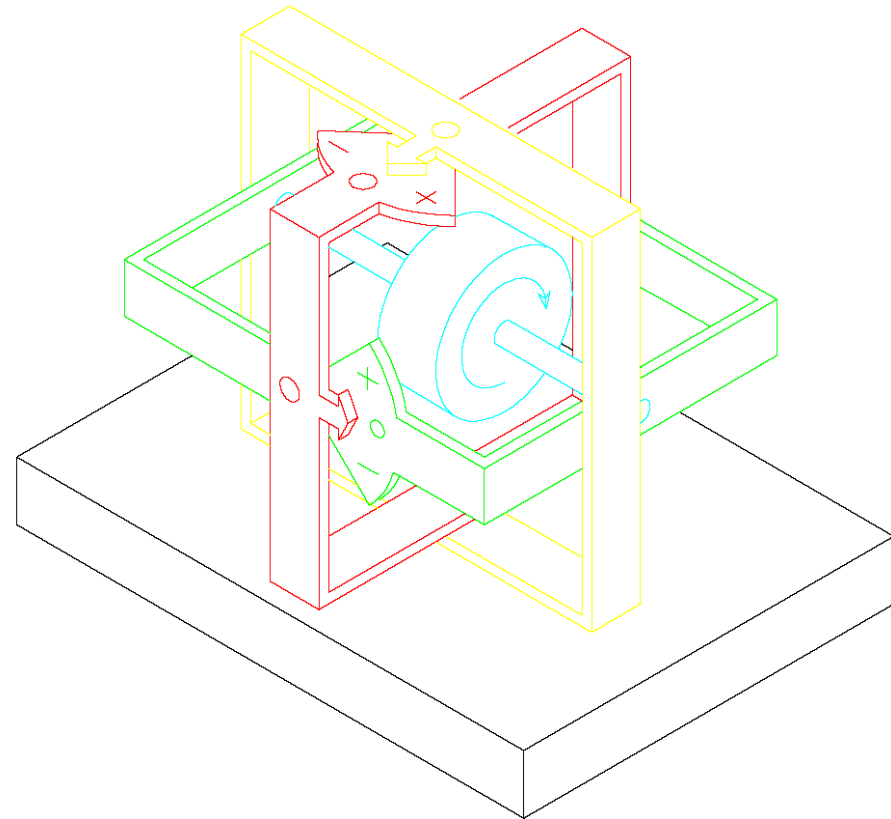
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    - 6.2.4.3 Gyroscopes
    - 6.2.4.1 Inertial Measurement Units
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## 6.2.4.3 Gyroscopes

- Word coined by Foucault in 1852.
- A spinning rotor
  - Rotor => has an axis of symmetry
  - Spinning => you know
- Two properties of interest
  - **Rigidity**: Angular momentum means hard to rotate
    - Not useful
  - **Precession**: Angular momentum means attempts to rotate one way cause rotations another way.



# 6.2.4.3.1 Mechanical Gyroscope

(Precession)

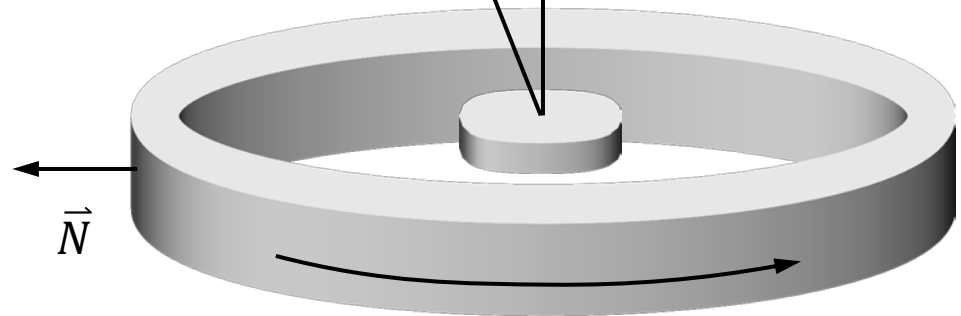
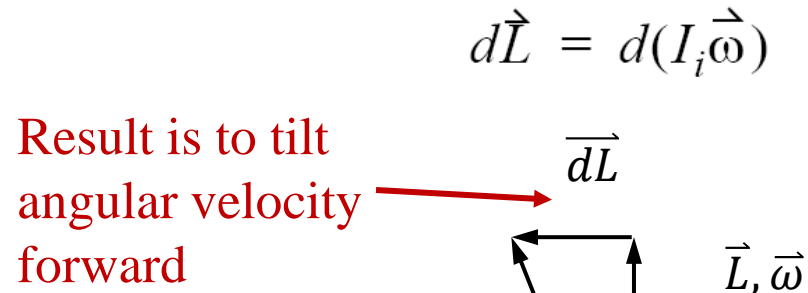
- Consider Euler's equation:

$$\vec{N} = \frac{d\vec{L}^i}{dt} = \frac{d(I^i \vec{\omega})}{dt}$$

- Inertia is diagonal so.

$$d\vec{\omega} \parallel \vec{N}$$

- Hence  $\omega$  (not the disk) must rotate in the direction of  $N$  !

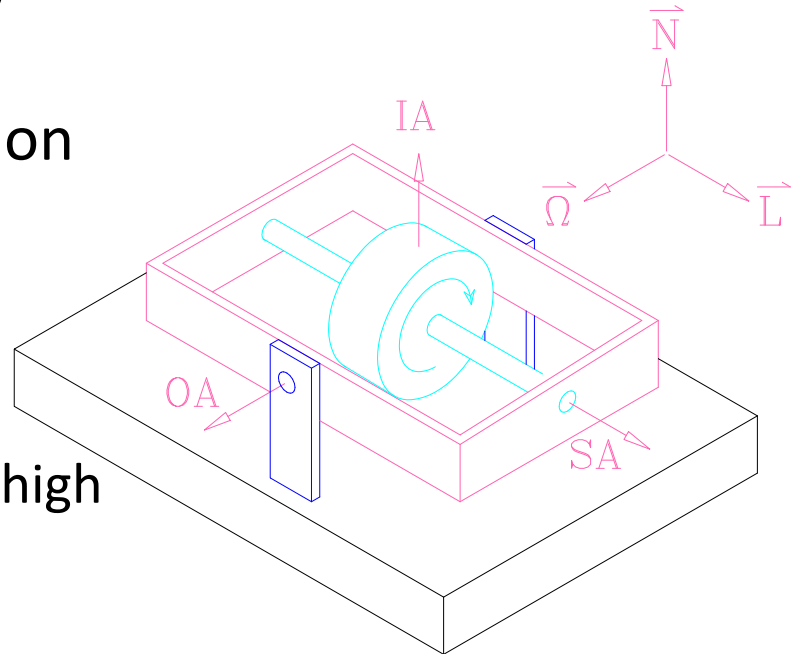


Torque is trying to lift the front up

# 6.2.4.3.1 Mechanical Gyroscope

## (Gyro Configurations)

- Some forms of drift are reduced by high  $\omega$  so rpms are usually very high.
- Rebalance loops used in precision devices.
- Packaging of one or more:
  - “rate” gyros: high spring, low damping (rate in  $\rightarrow$  rate out)
  - “integrating” gyros: low spring, high damp (angle in  $\rightarrow$  angle out)
  - “directional” gyros hold their azimuth, indicate heading (not gyrocompasses)
  - “vertical” gyros indicate pitch and roll (from gimbals) by staying vertical



## 6.2.4.3.3 Optical Gyroscopes

- Are **replacing mechanical ones** in most applications.
  - Used on all new civil transport aircraft (757, 767, Airbus 310)
  - **FOGs becoming common on robots.**
- Based on Sagnac Effect discovered in 1913.
  - Considered obscure at the time.
- Characteristics:
  - solid state or nearly so
  - More reliable, rugged
  - Good in harsh environments
  - Linearity of parts per billion
  - Large dynamic range

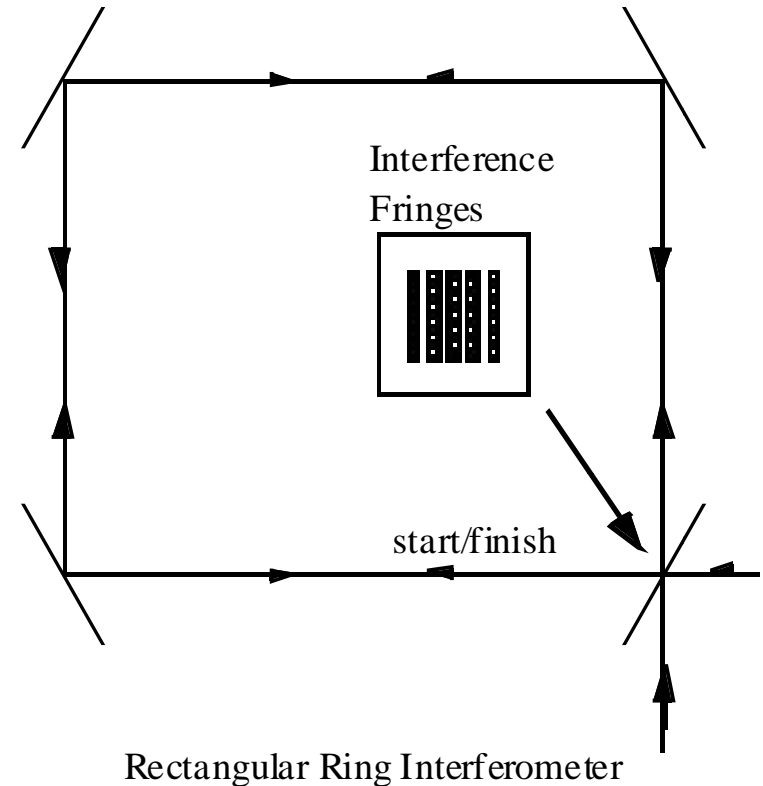
## 6.2.4.3.3 Optical Gyroscopes (Types)

- Ring Laser Gyros (RLGs)
  - Not solid state yet.
  - Becoming very popular
  - Will replace most inertial mechanical gyros
- Fiber Optic Gyros (FOGs)
  - Solid state
  - Newer to the market.
  - Will ultimately replace both mechanical and RLG's

# 6.2.4.3.3 Optical Gyroscopes

## (Sagnac Effect)

- While optical measurement of inertial translation is impossible (Michelson-Morely), optical measurement of inertial rotation is possible.
- Physical rotation of the ring interferometer gives rise to an apparent path length difference for two counter-rotating coherent beams of light.
- Effect can be understood in terms of:
  - different time of flight
  - different path length
  - different Doppler shifts



# 6.2.4.3.3 Optical Gyroscopes

(Sagnac Effect : Intuitive Derivation)

- Transit times for each beam in terms of itself:

$$t = s / v$$

$$t_+ = \frac{2\pi r + r\Omega t_+}{c} \quad t_- = \frac{2\pi r - r\Omega t_-}{c}$$

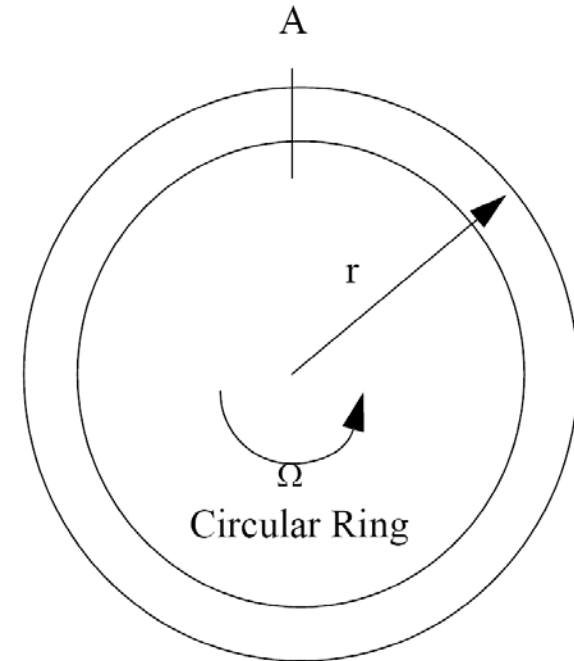
- Solve for times and take difference:

$$\Delta t = t_+ - t_- = \frac{4\pi r^2 \Omega}{c^2 - r^2 \Omega^2}$$

- For practical values of r:  $r\Omega^2 \ll c^2$   
so:

$$\Delta t = \frac{4\pi r^2 \Omega}{c^2}$$

$$c^2 = 10^{16}!$$



## 6.2.4.3.3 Optical Gyroscopes

(Sagnac Effect : Intuitive Derivation)

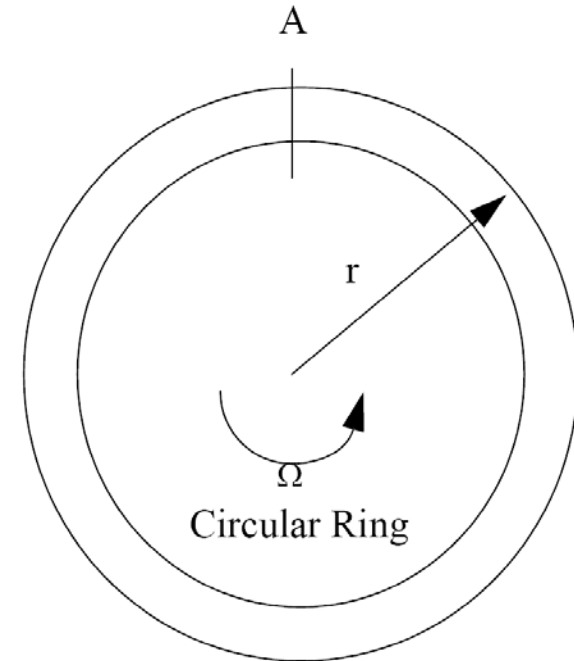
- So time difference is proportional to  $\Omega$ .

$$\Delta t = \frac{4\pi r^2 \Omega}{c^2}$$

- Alternatively, the path length difference is:

$$\Delta L = c\Delta t = \left(\frac{4\pi r^2}{c}\right)\Omega$$

- For rotation of 1 degree/sec and  $r = 0.1$  meter, the Sagnac effect gives 1/100 nanometer ( $1 \times 10^{-11}$  m) of deviation.





## 6.2.4.3.3 Optical Gyroscopes

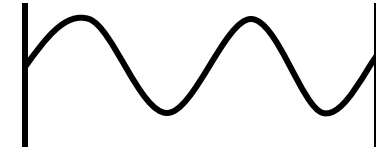
### (Sagnac Effect : Amplification)

- Two modern technologies **amplify** the Sagnac effect:
- Development #1: **lasers**
  - RLG converts Sagnac path length difference into a beat frequency.
- Development #2: **fiber optics**
  - FOG increases Sagnac path length by using several miles of fibre optic cable coiled into small package.

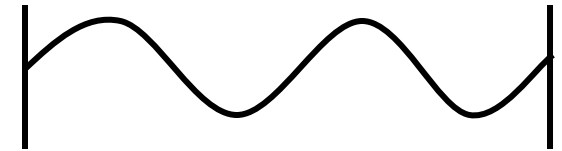
# 6.2.4.3.3 Optical Gyroscopes

## (Ring Laser Gyros)

- Place a lasing medium in the path of the light beams.
  - It's a laser and a gyro in one package.
- Cavity resonance condition determines the laser wavelength:
  - End up with two different frequency lasers when rotating.



Short cavity, short waves



Long cavity, long waves

# 6.2.4.3.3 Optical Gyroscopes (Ring Laser Gyros)

- Resonance condition:

$$m\lambda = L$$

- Resonant frequency:

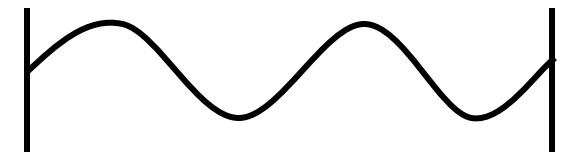
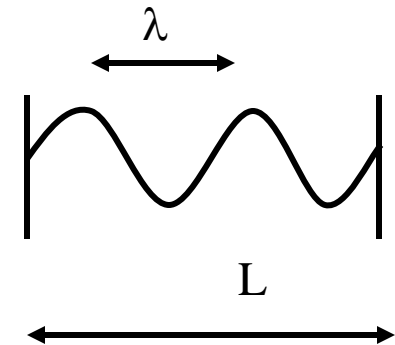
$$\omega = \frac{mc}{L}$$

- Frequency difference is now:

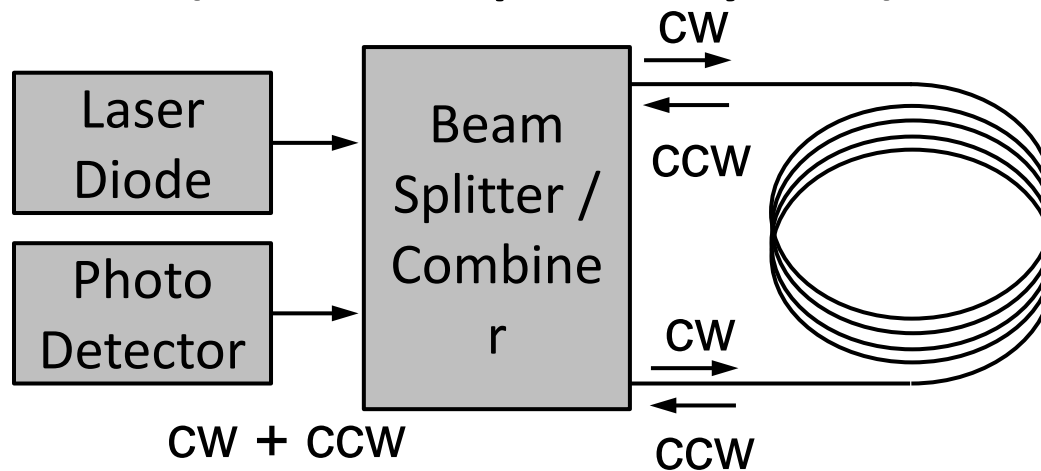
$$\Delta\omega = \omega_+ - \omega_- = \frac{mc}{L_+} - \frac{mc}{L_-} \approx \frac{mc\Delta L}{L^2} = \frac{\omega\Delta L}{L}$$

- Light frequencies  $\omega$  are in the  $5 \times 10^{14}$  range.

- Gives  $\Delta\omega$  of 50 KHz!
- Easy to measure beat frequency.



# 6.2.4.3.3 Optical Gyroscopes (Fiber Optic Gyros)

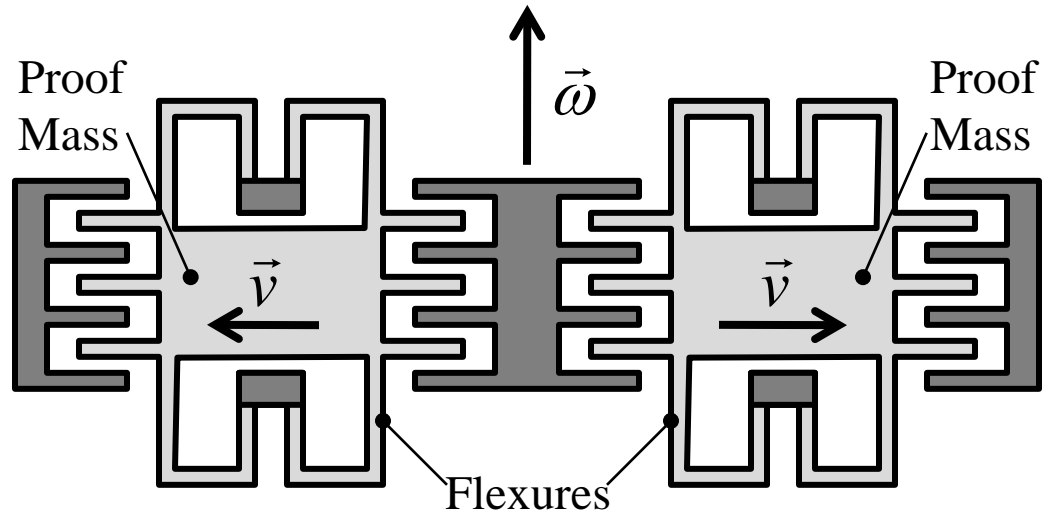


- Writing the earlier path length difference as a **phase difference**:

$$\Delta\phi = \frac{2\pi\Delta L}{\lambda} = \frac{8\pi^2 r^2 \Omega}{c\lambda}$$

- This is **multiplied by N** if we coil the fiber optic cable N times.
  - Simpler and more robust technology than RLGs but not yet equal in performance.

## 6.2.4.3.4 MEMS Gyros



- Cheap, and becoming useful.
- Tuning fork concept is common:
  - Light grey areas are driven to oscillate left and right.
  - They vibrate out of plane when device rotates around sensitive axis due to Coriolis force.

## 6.2.4.4. Inertial Measurement Units

- Typically 3 accels and 3 gyros (2 magnetometers) in one package.
- Price / Performance
  - 100 deg / hr = \$1K
  - 1 deg / hr = \$10K
  - 0.01 deg / hr = \$100K
- Single chip IMUs have just happened or are about to happen.

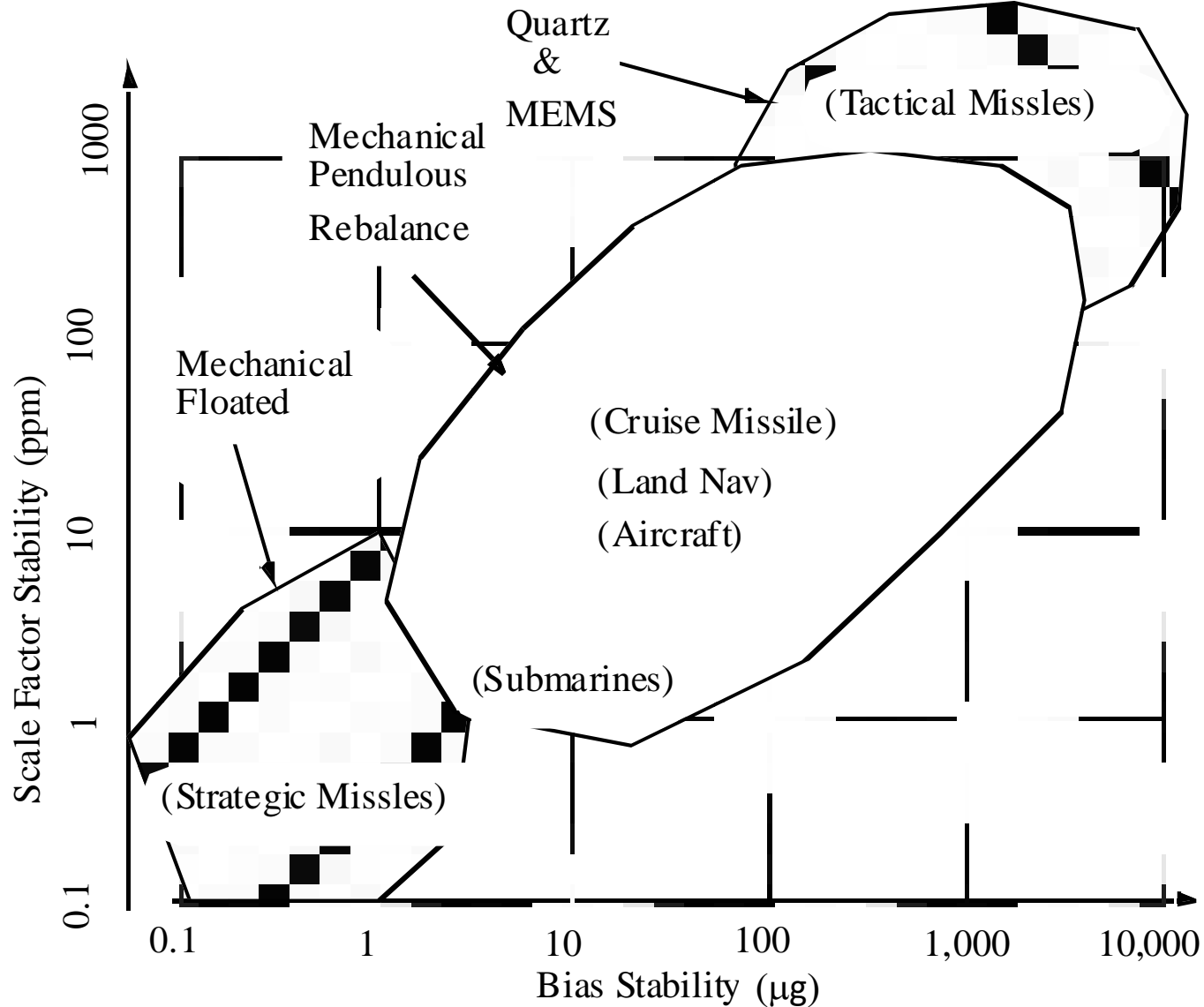


HG 1930  
Tactical Grade IMU



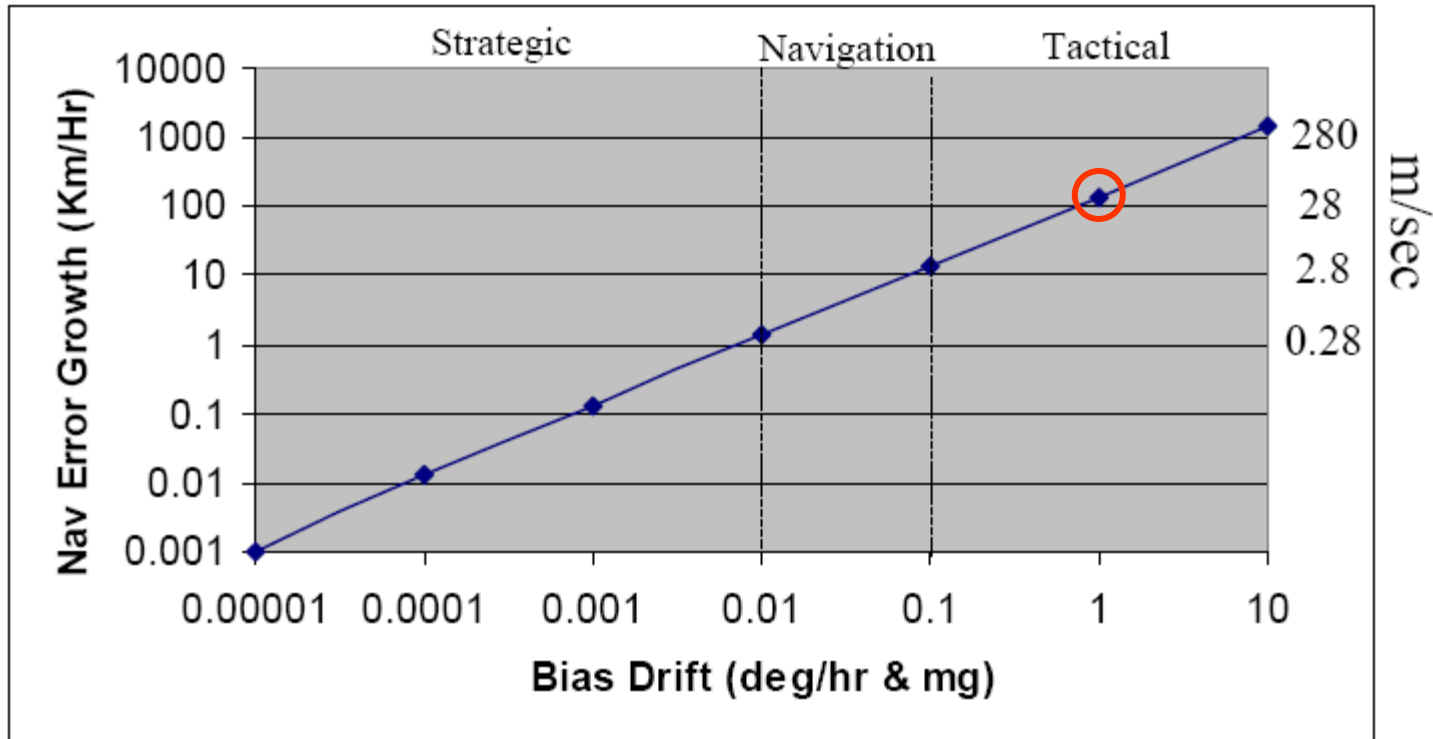
HG 9900  
Navigation Grade IMU

# Accelerometer Grades



# Impact on Nav Error

- 1 deg/hr attitude error rate is 60 mph (100 Km/hr) position error rate.



1 deg/hr = 60 nm / hr = 100 km/hr

$\frac{1}{2} a*t^2 = 64$  km after 1 hr

$a*t = 35$  m/sec after 1 hour



# Outline

- 6.2 Sensors for State Estimation
  - 6.2.1 Articulation Sensors
  - 6.2.2 Ambient Field Sensors
  - 6.2.3 Inertial Frames of Reference
  - 6.2.4 Inertial Sensors
  - Summary

# Summary

- Inertial frames are tricky to define.
- Accelerometers and inclinometers operate on similar principles
  - neither can distinguish gravitation from acceleration.
- Compasses indicate the local magnetic field.
  - not geographic north
  - Some error sources can be calibrated out.
- Gyros measure inertial rotation.
  - optical forms based on the Sagnac effect will be the ones used in the future. lasers and fiber optics made em practical.
  - MEMS devices are now very practical.